

# CRITICAL ITEMS LIST

ASSY NOMENCLATURE: EXHALATION VALVE

SYSTEM: CREW ESCAPE SYSTEM

REVISION

ASSY P/N: FTB33-3

SUBSYSTEM: LAUNCH ENTRY SLUT

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FMEA		NAME, QTY & DRAWING REF DESIGNATION	CRIT'Y	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	RATIONALE FOR ACCEPTANCE
REF	REV					
3.4.2		EXHALATION VALVE (2), 18951G-02	2/IR	<p>3.4.2 Mode: Valve fails closed</p> <p>Cause:                      • defective material                      • contamination</p>	Buildup of carbon dioxide if second valve fails	<p>1. DESIGN FEATURES TO MINIMIZE FAILURE MODE</p> <p>a. The exhalation valve is in current use by the Air Force.</p> <p>b. The valve is a mica disc.</p> <p>c. The case and seat is aluminum.</p> <p>d. The spring is phosphor bronze under calibrated compression.</p> <p>e. The valve opens at <math>1.65 \pm 0.15</math> inches H<sub>2</sub>O at a minimum input flow which shall not exceed 25 cc/minute.</p> <p>f. Resistance at flows of 0 to 95 slpm, 3.0 inches H<sub>2</sub>O maximum; 0 to 2 slpm, 0.3 inch H<sub>2</sub>O maximum above pressure setting.</p> <p>2. TEST OR ANALYSIS TO DETECT FAILURE MODE</p> <p>a. <u>Acceptance Testing</u></p> <p>(1) Flow of 25 cc/minute, at 70 psig - back pressure should read <math>1.65 \pm 0.15</math> inches H<sub>2</sub>O</p> <p>(2) Flow of 2 slpm, at 70 psig - back pressure should not increase more than 0.3 inch H<sub>2</sub>O.</p> <p>(3) Flow of 95 slpm, at 70 psig - back pressure should be less than 3.0 inches H<sub>2</sub>O</p> <p>b. <u>Certification Test</u></p> <p>(1) High altitude chamber Test, Brooks Air Force Base.</p> <p>(a) Unmanned testing series of gradual ascents and descents from ground level to 100,000 feet and rapid decompressions</p>

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SUPERSEDING DATE:

APPROVED BY: L. O. SCHLOSSER

DATE

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# CRITICAL ITEMS LIST

ASSY NOMENCLATURE: EXHALATION VALVE

SYSTEM: CREW ESCAPE SYSTEM

REVISION:

ASSY P/N: F1833-5

SUBSYSTEM: LAUNCH ENTRY SUIT

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FMEA		NAME, QTY & DRAWING REF DESIGNATION	CRITY	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	RATIONALE FOR ACCEPTANCE
REF	REV					
3.4.2		EXHALATION VALVE (2), 189516-02	2/18	<p>3.4.2 Mode: Valve fails closed</p> <p>Cause:                      • defective material                      • contamination</p>	Buildup of carbon dioxide if second valve fails	<p>(b) Manned test series</p> <ol style="list-style-type: none"> <li>1 Gradual ascents and descents to 100,000 feet.</li> <li>2 Rapid decompression to 90,000 feet</li> <li>3 Endurance runs rapid decompression to 100,000 feet for 37 minutes.</li> </ol> <p>(2) Live jumped at Naval Weapons Center.</p> <ol style="list-style-type: none"> <li>(a) At 200 knots, 25,000 feet, four jumps.</li> <li>(b) At 110 knots, 10,000 feet, four jumps.</li> <li>(c) At 110 knots, 6,000 feet, four jumps.</li> <li>(d) At 170 knots, 15,000 feet, four jumps.</li> <li>(e) At 185 knots, 20,000 feet, four jumps.</li> <li>(f) Water drop at 30 feet per second (fps), two jumps.</li> <li>(g) Water drop at 27 fps, two jumps.</li> </ol> <p>c. <u>Turnaround Test</u> (In accordance with PIA 23033)</p> <ol style="list-style-type: none"> <li>(1) Flow of 25 cc/minute, at 70 psig - back pressure should read <math>1.65 \pm 0.15</math> inches H<sub>2</sub>O</li> <li>(2) Flow of 2 slpm, at 70 psig - back pressure should not increase more than 0.3 inch H<sub>2</sub>O.</li> <li>(3) Flow of 95 slpm, at 70 psig - back pressure should be less than 3.0 inches H<sub>2</sub>O.</li> </ol> <p>3. INSPECTION</p> <ol style="list-style-type: none"> <li>a Visual inspection of parts for defects.</li> <li>b One hundred percent visual inspection during assembly</li> <li>c Visual inspection on glyptal seal for defect</li> <li>d Visual inspection for contamination.</li> </ol>

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# CRITICAL ITEMS LIST

ASSY NOMENCLATURE: EXHALATION VALVE

SYSTEM: CREW ESCAPE SYSTEM

REVISION

ASSY P/N: F1033-5

SUBSYSTEM: LAUNCH ENTRY SUIT

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FMEA		NAME, QTY & DRAWING REF DESIGNATION	CRITY	FAILURE MODE AND CAUSE	FAILURE EFFECT ON END ITEM	RATIONALE FOR ACCEPTANCE
REF	REV					
3.4.1		EXHALATION VALVE (2), 18951G-02	2/1R	<p>3.4.2 Mode: Valve fails closed</p> <p>Cause:</p> <ul style="list-style-type: none"> <li>• defective material</li> <li>• contamination</li> </ul>	Buildup of carbon dioxide if second valve fails	<p>e. Verify flows are within specifications of the acceptance test</p> <p>f. Verify exhalation valve is cleaned to level 300 in accordance with JSCM 5322</p> <p><u>Turnaround Inspection:</u> In accordance with PIA 23033)</p> <p>a. Visual inspection of parts for defects.</p> <p>b. One hundred percent visual inspection during assembly.</p> <p>c. Visual inspection on glyptal seal for defect.</p> <p>d. Visual inspection for contamination.</p> <p>e. Verify flows are within specifications of the acceptance test.</p> <p>f. Verify exhalation valve is cleaned to level 300 in accordance with JSCM 5322.</p> <p>4. FAILURE HISTORY None. This exhalation valve is used by the Air Force in high altitude suits for high performance aircraft and Dryden Flight Research Center.</p> <p>5. OPERATIONAL USE</p> <p>a. Operational Effect of Failure - Possible loss of crewmember if both valves fail.</p> <p>b. Crew Action - None</p> <p>c. Crew Training - Not applicable</p> <p>d. Mission Constraints - None.</p> <p>e. In-Flight Checkout - None. Crew could inspect anti-suffocation valve, but could not repair or replace a defective valve.</p>

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